In-flight upset, Boeing 747-236, G-BDXL, February 27, 2000

Micro-summary: This Boeing 747 encountered an in-flight upset during descent.

Event Date: 2000-02-27 at 2100 EST

Investigative Body: National Transportation Safety Board (NTSB), USA

Investigative Body's Web Site: http://www.ntsb.gov/

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1. Accident reports can be and sometimes are revised. Be sure to consult the investigative agency for the latest version before basing anything significant on content (e.g., thesis, research, etc).

2. Readers are advised that each report is a glimpse of events at specific points in time. While broad themes permeate the causal events leading up to crashes, and we can learn from those, the specific regulatory and technological environments can and do change. Your company's flight operations manual is the final authority as to the safe operation of your aircraft!

3. Reports may or may not represent reality. Many many non-scientific factors go into an investigation, including the magnitude of the event, the experience of the investigator, the political climate, relationship with the regulatory authority, technological and recovery capabilities, etc. It is recommended that the reader review all reports analytically. Even a "bad" report can be a very useful launching point for learning.

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National Transportation Safety Board	1	NTSB ID:	NYC00LA08	5	Aircraft Registration Number: GBDXL					
FACTUAL REPORT		Occurrenc	e Date: 02/27	7/2000	Most Critical Ir	Most Critical Injury: Serious				
AVIATION ETYBON	(Occurrence	e Type: Accio	lent	Investigated B	Investigated By: NTSB				
Location/Time	-									
Nearest City/Place	State	e Zip Code Local Time			Time Zone					
PROVIDENCE	RI	02	901	2100	EST					
Airport Proximity: Off Airport/Airstrip	Distanc	e From La	nding Facility:	sirport:						
Aircraft Information Summary										
Aircraft Manufacturer			Model/Serie	S			Type of Aircraft			
Boeing			747-236				Airplane			
Sightseeing Flight: No		Ai	r Medical Tı	ansport Flight:	No					
Narrative										
On February 27, 2000, abor British Airways, PLC., as fi vicinity of Providence, Rhod passengers were not injured. flight attendant sustained mi instrument flight rules fligi Airport (LHR), England, Unite (JFK), Jamaica, New York. The si In an interview with a Feder that the fasten seat belt sign Category III (CAT III) land time, the flight engineer be (CAT I) landing, due to an imj "number one bus-tie-breaker," 5-degrees nose-up. The airj The pilot disconnected the si continued a normal descent. the pilot reported that the pip the effect was very similar to airplane's electrical system suj Examination of the airplane's si JFK did not reveal any discrepan The accident airplane was equi quick access recorder (OQAR) provided to the Safety Board electrical discontinuity around	ut 210 light e Isl on ht pl ed Ki schedu ral Av was " ing, egan t provem the plane autopi tch-up the e pply c autopi ncies. ipped . The t, on t	One passenger received serious injures, while 10 passengers and r injures. Instrument meteorological conditions prevailed and plan had been filed for the flight that departed London-Heathr Kingdom, destined for the John F. Kennedy International Airpo eduled international flight was conducted under 14 CFR Part 129. Aviation Administration (FAA) Inspector, the flight crew report s "off," and the airplane's electrical system was configured for , when they began a descent from "flight level 350." At the sa n to reconfigure the airplane's electrical system for a Category vement in landing visibility. When the flight engineer closed t he airplane's pitch changed from 2-degrees nose-down, to abo ne was utilizing the "A" autopilot system, which remained engage opilot, leveled the airplane, re-engaged the autopilot, and th he airplane landed at JFK without further incident. Additionall -up was accompanied by numerous momentary instrument failures, a e electrical changeover that is experienced on the ground when t y changes from ground power to aircraft power. opilot and electrical system performed by maintenance personnel es. ed with a Penny & Giles flight data recorder (FDR), and an optic The data from the recorders was downloaded by British Airways, a Examination of both the FDR and the OQAR information revealed								
using the airplane's "B" autopilot system, and closing the "number two bus-tie-breaker." Additionally, the flight crew reported that the airplane "felt light in pitch." A subsequent inspection of the airplane revealed that the number 1 and 2, "elevator feel computer" pitot connections were capped. Review of the airplane's maintenance history revealed that the airplane underwent an "inter 2 check" at a British Airways maintenance facility between February 5							or feel computer" revealed that the			

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Narrative (Continued)									
and 23, 2000. According to a British Airways quality inspection report, during the time of the inter 2 check, the pitot connections to the elevator feel computer were disconnected by maintenance personnel in order to perform pitot static system checks "in-accordance-with (IAW) the [airplane] Maintenance Manual [Chapter] 34-11-00." A functional check of the feel computer was not performed before the airplane was returned to service.									
Review of the Boeing basic 747 Maintenance Manual Chapter 34-11-00, Pitot-Static Adjustment/Test, revealed an "Elevator Feel Light Test" and the following note:									
	"The following test must be performed to ensure that auxiliary pitot systems No. 1 and 2, which were disconnected prior to system leakage test, are properly reconnected"								
British Airways utilized a customized version of the 747 Maintenance Manual, which was provided by Boeing. Review of the maintenance manual chapter 34-11-00, Pitot-Static Adjustment/Test revealed that the customized section did not contain the requirement for an elevator feel light test. The section did specify that a "leak check" be performed after the pitot-static lines are reconnected. A representative from Boeing stated that if the pitot-static connections to the elevator feel computer were left disconnected and capped, then a "leak check" would not identify an unconnected elevator feel computer, provided that the caps were pressure tight.									
The Boeing representative also stated that Boeing intends to publish a revised customized 747 Maintenance Manual for British Airways, which will include an elevator feel light test in Chapter 34-11-00. Additionally, Boeing will revise the customized maintenance manuals for four other 747 operators.									
Boeing provided information on the computer.	ne effect of disconnected pitot-static lines on the elevator feel								
According to Boeing, the elevator control system required artificial feel forces that were provided by a combination of mechanical and hydraulic springs contained in the feel unit. The feel computer programs hydraulic pressure to the feel unit actuators as a function of pitot pressure and stabilizer position. With the lines disconnected, the feel computer would react as if the airspeed is low and thus the feel unit forces would be less than expected. The autopilot reacts against artificial feel forces to regulate the deflection of the elevator surface. If the artificial feel forces were low, the autopilot command would cause greater than normal elevator deflection, resulting in a larger upset of the airplane than would normally be encountered.									
time of the accident should have	Boeing calculated that the normal autopilot elevator authority for the flight conditions at the time of the accident should have been about 4 degrees. The estimated actual elevator deflection during the accident sequence was 6.87 degrees nose up, and 6.97 degrees nose down.								
British Airways reported they were able to duplicate the pitch-up during two test flights, using two other 747-200 airplanes; however, the magnitude of the elevator movement experienced during the test flights remained within the autopilot elevator authority.									
Subsequently, British Airways inst procedure:									
selecting CLOSE on the BTB. The a	ry to CLOSE a BTB, the autopilot must be disconnected prior to autopilot may be reselected once normal conditions are confirmed. possible short term flight instrument failures may occur."								
The Boeing Operations Manual for the	e airplane, stated:								
"When the No. 1 and No. 2 Bus Tie	"When the No. 1 and No. 2 Bus Tie OPEN lights illuminate due to triple channel operation, reclose								

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Narrative (Continued)		I						
BTBs during accomplishment of the A	FTER LANDING PROCEDURE or when i	n stabilized flight."						
The Boeing Flight Crew Training included the note:	Manual for the airplane, Au	tomatic Flight, Go-Around section						
"The automatic bus isolation sys disengages, however, bus tie break do not have auto-paralleling cir phase relationship. Closing of cause a voltage fluctuation. momentary airplane instrumentation	ers 1 and 2 will not reclose aut cuits and when placed to CLOSE w the bus tie breakers during ce While these voltage fluctuati	comatically. The bus tie breakers will connect the bus regardless of ertain out of phase conditions may						
At the time of the accident, modified by a supplemental type (FMC) interface with the autopilot data adapter unit. The modifi According to Boeing and Honeywell, this configuration.	certificate to allow for a mo . The modification incorporated ed British Airways 747-200 airp	odified flight management computer l a Honeywell FMC, and a Honeywell planes had been in use since 1984.						
British Airways reported they were in the process of "retiring" their 747-200 airplanes and expect to have all of their 747-200 airplanes out of service by April of 2002. The airplanes are being sold to a leasing company that intends to utilize the airplanes as freighters.								
Thirty two operators of "classic flight crews used when they rec also asked to report any uncom reconfigurations. Twenty-one opera	onfigured from a CAT III approac manded aircraft motions as a	h to a CAT I approach. They were						
Of the twenty one operators wh approaches, of which, 4 operato and 2 operators published proce respondents reported uncommanded ai	rs stated they waited until afte dures for closing the BTBs afte	er landing to reconfigure the BTBs er a missed approach. None of the						
The source of the pitch-up comman test flights, was not determined pitch-up characteristics were not o	; however, when the autopilot sy	stem was properly configured, the						

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FACTUAL REPORT		urrenc	rence Date: 02/27/2000									
AVIATION	Occ	urrenc	nce Type: Accident									
Landing Facility/Approach Informa			-) 1 -									
Airport Name	Airpo	ort ID:	Airport Eleva	tion	Run	way Used	Runwa	ay Length	h Ru	nway Width		
				Ft	. MSL	0						
Runway Surface Type:	Runway Surface Type:											
Runway Surface Condition:												
Type Instrument Approach:												
VFR Approach/Landing:												
Aircraft Information												
Aircraft Manufacturer Boeing			Model/ 747-2						Serial N 22305	l Number 05		
Airworthiness Certificate(s): Transport												
Landing Gear Type: Retractable - Tricy	cle											
Homebuilt Aircraft? No Numb	umber of Seats: 394 Certified Max Gross Wt. 820000 LBS Number of Engines: 4							es: 4				
Engine Type: Turbo Fan		Engine Manufacturer:Model/Series:Rolls-RoyceRB211-524D4							Rated Power: 52810 LBS			
- Aircraft Inspection Information												
Type of Last Inspection			Time Since Last Inspection						Airframe -	Total Time		
Continuous Airworthiness		02/	/2000					59 Ho	ours		648 Hours	
- Emergency Locator Transmitter (ELT) Ir	nformation											
ELT Installed? No	ELT Operated?				ELT	Aided i	n Locating Ac	cident S	Site?			
Owner/Operator Information												
Registered Aircraft Owner			Street A	ddress P.O. BO	X 365							
BRITISH AIRWAYS, PLC.		City							State OF	Zip Code		
			Street A				<u>.</u>		I	01		
Operator of Aircraft		Same as Reg'd Aircraft Owner								1		
Same as Reg'd Aircraft Owner		City State Zip Co							Zip Code			
Operator Does Business As: Operator Designator Code:												
- Type of U.S. Certificate(s) Held: None												
Air Carrier Operating Certificate(s):												
Operating Certificate:	Operating Certificate: Operator Certificate:											
Regulation Flight Conducted Under: Part 129: Foreign												
Type of Flight Operation Conducted: Sch	eduled; Internatio	onal; F	Passen	ger/Cargo								
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	ACTUAL RI	17		Occurren	Occurrence Date: 02/27/2000									
	AVIATI	~ ~			Occurrence Type: Accident									
	First Pilot Information City State Date of Birth Age													
												Date of	Dirti	
On File	1	On File 47										47		
Sex: M Seat Occupied: Left Principal Profession: Civilian Pilot Certificate Number:														
Certificate(s): Airline Transport														
Airplane R	Rating(s): Multi	i-engine Lai	nd											
Rotorcraft/	/Glider/LTA: None	Э												
Instrument	t Rating(s): Airpl	ane												
Instructor	Rating(s): Airpl	ane Multi-e	ngine; Instr	ument Airpl	ane									
Type Ratin	ng/Endorsement fo	or Accident/Ir	cident Aircra	aft? Ves				Current F	Biennial Flig	nht R	eview?			
	ert.: Class 1			s: Valid Me	dicalw/ w						st Medical	Exam: 0	0/1000	
		Weak							Date				3/1333	
					1									
- Flight Tir	me Matrix	All A/C	This Make and Model	Airplane Single Engine	Airplane Mult-Engine	Ni	ght	Actual	Instrument Simu	strument Simulated		t	Glider	Lighter Than Air
Total Time	e	13200												
Pilot In Co	ommand(PIC)	4600	2670								_			
Instructor			800			_								
Last 90 Da	-		94			_								
Last 30 Da			56			_								
Last 24 Ho														
Seatbelt U	Ised? Yes	Shou	Ider Harnes	s Used? Yes	5		IOXICO	ology Pe	erformed?	No		Second P	vilot? Yes	3
	an/Itinerary													
	ight Plan Filed: IF	R												
Departure										Airport Identifier		Departure Time		Time Zone
LONDON	N						OF		LHR		183	0	1	GMT
Destination	n						State	Airport Ide	ntifie	r				
JAMAIC	A						NY		JFK					
Type of Clearance: IFR Type of Airspace: Class A														
	•	A												
	Information													
Source of Briefing: Company														
		~.',												
Method of	f Briefing:													
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	ACTUAL REPOP		Occ	urrence Date	: 02/27/2	000		1					
	Z AVIATION ETYBON			Occurrence Type: Accident									
Weather	Information												
WOF ID	Observation Time	Time Zone	WOFF	Elevation	WOF D	stance From	Accie	dent Site		Direction From A	Accident Site		
							, 10011						
	0000			0 Ft. MSL				0 NM			0 Deg. Mag.		
Sky/Lowes	st Cloud Condition: Unk	nown				0 Ft. AGI	L	Condition of	Condition of Light: Night/Dark				
Lowest Ce	eiling: Unknown			0 Ft. AGL	Visib	ility:	0	SM	Alti	meter:	"Hg		
Temperatu	ure: °C	Dew Point:		°C Win	d Direction:				Dei	nsity Altitude:	Ft.		
Wind Spee	ed:	Gusts:		Wea	ther Condt	ions at Accid	ent Si	^{ite:} Instrum	ent C	Conditions			
Visibility (F	RVR): 0 Ft.	Visibility	(RVV)	0 SM	Intensit	y of Precipita	tion: I	Unknown					
Restriction	ns to Visibility:												
	-												
Type of Pr	ecipitation:												
1900111	colpitation												
Accident	Information												
Aircraft Da	mage: None		Aircra	aft Fire: Nor	e			Aircraft Exp	olosio	n None			
Classificat	ion: Foreign Registere	ed/U.S. Soil	I										
- Injury Su	mmary Matrix	Fatal	Serious	Minor	None	TOTAL							
First Pi					1	1							
Secon	d Pilot				1	1							
Studer	nt Pilot												
Flight I	Instructor												
Check	Pilot												
Flight E	Engineer				1	1							
Cabin /	Attendants			1	14	15							
Other (Crew												
Passer	ngers		1	10	354	365							
- TOTAL /	ABOARD -		1	11	371	383							
Other 0	Ground	0	0	0		0							
- GRANI	D TOTAL -	0	1	11	371	383							
			FACT	UAL REP	ORT - AV	IATION					Page 4		

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FACTUAL REPORT	Occurrence Date: 02/27/2000	
AY IATION	Occurrence Type: Accident	
Administrative Information		
Investigator-In-Charge (IIC)		
LUKE SCHIADA		
Additional Persons Participating in This Accident/Incid	ent Investigation:	
MIKE CARTELLI FAA FSDO GARDEN CITY, NY		
SIMON LIE BOEING SEATTLE, WA		
MARTIN BUZZARD BRITISH AIRWAYS LONDON, UK		
GREG NIECIECKI HONEYWELL PHOENIX, AZ		