Rejected takeoff after engine failure, Boeing 727-212, C-FRYS, March 30, 1998

Micro-summary: This Boeing 727 experienced a failure of the #2 engine.

Event Date: 1998-03-30 at 2114 EST

Investigative Body: National Transportation Safety Board (NTSB), USA

Investigative Body's Web Site: http://www.ntsb.gov/

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NTSB ID: MIA98FA112

Aircraft Registration Number: CFRYS

Occurrence Date: 03/30/1998

Most Critical Injury: Serious

Occurrence Type: Accident

Investigated By: NTSB

Location/Time

Airport Proximity: On Airport	Distance From	m Landing Facility:	-	Direction Fro	m Airport:
FORT LAUDERDALE	FL	33315	2114	EST	
Nearest City/Place	State	Zip Code	Local Time	Time Zone	

Aircraft Information Summary

· · · · · · · · · · · · · · · · ·		
Aircraft Manufacturer	Model/Series	Type of Aircraft
Boeing	727-212	Airplane

Sightseeing Flight: No

Air Medical Transport Flight: No

Narrative

Brief narrative statement of facts, conditions and circumstances pertinent to the accident/incident:

HISTORY OF THE FLIGHT

On March 30, 1998, about 2114 eastern standard time, a Boeing 727-212, Canadian registration C-FRYS, registered to and operated by Royal Aviation, Inc., as flight 311, a Title 14 CFR Part 129, nonscheduled international passenger flight, from Fort Lauderdale, Florida to Toronto, Canada, experienced a failure of No. 2 engine during takeoff roll at Fort Lauderdale-Hollywood International Airport. Visual meteorological conditions prevailed and an instrument flight plan was filed. The airline transport-rated pilot, first officer, flight engineer, 5 flight attendants, and 169 passengers, including 2 lap children, were not injured. Three passengers sustained serious injuries and 14 passengers sustained minor injuries during the emergency evacuation after takeoff was aborted. The flight was originating at the time of the accident.

The pilot stated that during takeoff roll from runway 9L, between 90 and 100 knots indicated airspeed, he felt a "thud" and observed an "engine fail" light and No. 2 engine fire warning . At about 110 knots, he aborted the takeoff, brought the airplane to a stop on a high-speed taxiway, and requested that the "engine fire on ground" and "aborted takeoff" check-lists be read and accomplished. When the tower advised that smoke was still emanating from the engine tailpipe, the pilot commanded that a passenger emergency evacuation be accomplished, and that the "emergency evacuation" checklist be read and accomplished. The four fuselage doors and slides were used, as well as the four overwing exits. The aft airstairs were not used. The pilot further stated that the flight attendant's safety brief, including emergency evacuation exits and escape routes had been given to the passengers, according to company procedures. The evacuation took about 3 minutes.

Fire rescue personnel stated that upon arrival at the aircraft they observed that some passengers had assembled on-wing at the wing-tips. They were directed back toward the wing root area by the first officer and fire rescue personnel on the ground, and that no one actually jumped from the wing tip area.

PERSONNEL INFORMATION

Information on the pilot, first officer, and flight engineer is included in this report under First Pilot, in supplement E, and in the attached Pilot/Operator Aircraft Accident Report.

AIRPLANE INFORMATION

The aircraft, a Boeing 727-212, had been modified in accordance with Federal Aviation Administration Supplement Type Certificate SA 5938 NM. As part of the modification, the wing-tips had been modified to the winglet type. The trailing edge flaps were limited to 30 degrees from the

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Narrative (Continued)

original 40 degrees, by a blocking device installed on the cockpit flap quadrant. Examination of the aircraft after the accident showed the ground evacuation emergency checklist mounted on the cockpit yoke, and the Royal Airlines printed cockpit checklist called for "FLAPS-40", which was not possible. Examination of the aircraft also showed the passenger seatback pockets contained the Royal Airlines, Boeing 727-200 Safety Card, that depicts the escape route from each emergency escape exit. For additional information, see attached flight manual supplement for Boeing 727-200 series with STC SA 5938 installed under, Other Pertinent Forms and Reports. A copy of the seatback safety card is attached under, Other Pertinent Forms and Reports.

METEOROLOGICAL INFORMATION

Visual meteorological conditions prevailed at the time of the accident. For additional information, see Weather Information included in this report.

FLIGHT RECORDERS

The digital flight data recorder was removed from the airplane on April 1, 1998, and sent to the NTSB Vehicle Recorders Laboratory, Washington, D.C. The data extracted confirms the airplane's positioning to runway heading, the 3-engine spool-up to about 1.6 EPR, (engine pressure ratio, the primary thrust setting instrument) for 3 seconds, and the continuous increase of thrust to takeoff EPR of 2.1. The number 2 engine EPR sharply decreases to about 1.0 at an airspeed of 105 knots, followed immediately by the sharp decrease and slight increase of EPR's 1 and 3. (For additional information see the Flight Data Recorder/Specialist's Factual Report of Investigation.)

WRECKAGE INFORMATION

Postaccident inspection of the aircraft was conducted by NTSB and FAA personnel on April 1, 1998. The No.2 engine failure was uncontained just aft of the low pressure turbine, (LPT). The engine was cut completely in half just aft of the LPT. The fan exhaust duct was completely severed and the LPT case was completely separated from the remainder of the turbine exhaust case. The LPT containment shield was broken into several pieces. The LPT disks were intact but the blades were all fractured just above their root platforms, and most of them had been liberated. The two fan rotors and the 3rd stage compressor rotor could be rotated as a unit, but rotated independently of the remainder of the low pressure rotor assembly. The aft cowling halves had fire blistered paint and numerous shrapnel perforations just aft of the LPT's plane of rotation. Airframe and engine damage is included on page 3 and page 5 of the attached Powerplants Group Chairman's Factual Report.

Teardown inspection of the No. 2 engine by the Powerplants Group showed initial engine component failure was one or more of the 12 low pressure compressor, (LPC) rear tierods. One tierod had failed at its 4th stage compressor disk land and the 11 others failed adjacent to their threaded ends. (For additional information see the Powerplants Group Chairman's Factual Report.)

MEDICAL INFORMATION

A report from on-scene EMS personnel indicated that 17 passengers were treated at the scene and transported to three area medical centers for further evaluation. Injuries to those passengers included one small arm laceration, one hip trauma, one hypertensive reaction, and 11 with ankle trauma. The serious injuries were three fractured ankles sustained while sliding off the wing flap trailing edge to the ground.

TESTS AND RESEARCH

The 12 LPC rear tierods, (PN 789550) and pieces of the failed low pressure turbine (LPT) containment shield were removed and shipped to the NTSB Materials Laboratory, Washington, D. C. It

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Narrative (Continued)

was revealed that an anti-gallant treatment to the tierods that are used to bolt the 3rd through 6th compressor disks together as a unit was not present. The treatment is applied to the lands of the tierods at manufacture as specified in Pratt & Whitney Service Bulletin No. 5407. One tierod was fractured transversely across the tierod's 4th stage compressor disk land. The entire fracture face on the tierod had surface corrosion, suggesting that the tierod had been broken for some time. The 11 remaining LPC rear tierods were fractured at their forward ends between the 3rd stage compressor disk lands and the tierod nut threads. These 11 fractures appeared to be overload caused. Pockmarks and bending damage suggest that after initial tierod fracture and liberation of its tierod end/nut, it tumbled around in the cavity at the front of the LPC and impacted the other tierod ends/nuts until they eventually fractured, causing the catastrophic compressor separation. Eleven equally spaced fan exit vanes showed imprints of the tierod ends/nuts on their leading edges, indicating that the 11 remaining tierods fractured nearly simultaneously at engine failure. For additional information see attached, NTSB Poweplants Group Chairman Factual Report and NTSB Materials Laboratory Factual Report.

Regarding the discrepancy between printed checklist response of "Flaps 40" for an emergency evacuation flap setting and having the maximum flap setting physically blocked at "Flaps 30", Royal Airlines flight operations personnel stated that the checklists simply had not been updated. Because of the flap extension limitation, the difference between 30 degrees and 40 degrees was measured. Standing on the wing's trailing edge, the difference to an evacuating passenger is visually imperceptible due to the fact that after 30 degrees, additional flap movement is straight down, relative to the aircraft's longitudinal axis. Measured from the ground to the trailing edge of the flap, at 40 degrees, inboard flap is 27 inches, inboard end, to 33 inches, outboard end. At 30 degrees, inboard flap is 42 inches, inboard end, to 47 inches, outboard end.

The Boeing Aircraft Company stated that the airplane was designed to the FAA requirements for overwing egress. Design considerations when authorizing flaps 30 as a maximum is within compliance with FAR Part 25.810 which requires that only heights of greater than six feet, (from passenger standing surface to ground level) require additional egress assistance be provided. The Boeing statement is located under, Reports from Parties to the Investigation.

The Royal Airlines Flight Attendant Manual's recommended pre-departure safety announcement states the number of exits, their locations, how to identify them, and which ones are equipped with inflatable slides. The announcement makes no mention of the escape path direction once the evacuee steps through the overwing exit and out onto the wing. Excerpts from the manual are located under, Other Pertinent Forms and Reports.

ADDITIONAL INFORMATION

The aircraft was released by NTSB to the airline's contract ground handler at about 2309 on March 31, 1998. The deployed emergency evacuation slides were removed and the aircraft was towed to a terminal gate for passenger personal effects retrieval. The aircraft was subsequently towed to a hangar ramp area where investigators examined the airplane the next morning.

Except for the engine LPC rear tierods and the LPT containment shield pieces, the engine was released to Royal Airlines at the completion of the teardown inspection on April 8, 1998. The tierods were returned to Royal Airlines on August 12, 1998. The pieces of the LPT containment shield were discarded per authorization from Royal Airlines on August 12, 1998. The FDR and CVR were returned to Royal Airlines on December 9, 1998.

Additional parties to the NTSB investigation not listed in the factual portion of the report were Dennis Campbell, Transport, Canada, and Pamela Rosnik, Boeing Aircraft Co.

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FACIDAL REPORT	urreno	Terice Date: 03/30/1998											
AVIATION	Occ	urren	се Туре:	Accident									
Landing Facility/Approach Information	n												
Airport Name	Airpo	ort ID:	Airport El	evation	Rur	way Used	Runwa	nway Length		Runv	vay Width		
FT. LAUDERDALE/HOLLYWOOD		FLL	-	11	Ft. MSI	_ 9L		9001			150		
Runway Surface Type: Asphalt													
Runway Surface Condition: Dry													
Type Instrument Approach:													
VFR Approach/Landing:													
Aircraft Information													
Aircraft Manufacturer Boeing			Model/ 727-2						Serial 2134	Numbe	er		
Airworthiness Certificate(s): Transport													
Landing Gear Type: Retractable - Tricycle	е												
Homebuilt Aircraft? No Number	of Seats: 195		Certified	d Max Gros	s Wt.		19920) LBS	Numbe	er of Er	r of Engines: 3		
Engine Type: Turbo Fan					Engine Manufacturer: Model/Series: JT8D-17A						Rated Power: 16000 LBS		
- Aircraft Inspection Information													
Type of Last Inspection		Dat	Date of Last Inspection Time Since				nce Last Ins	ce Last Inspection			me To	tal Time	
Continuous Airworthiness		03	03/1998 93 Ho					ours 54355 Hours					
- Emergency Locator Transmitter (ELT) Info	rmation												
ELT Installed? Yes E	LT Operated? N	0			EL	T Aided i	in Locating A	ccident S	Site?				
Owner/Operator Information													
Registered Aircraft Owner			Street A		RAHAI	M STUA	RT BLVD.	N					
ROYAL AVIATION, INC			City DORVAL								te	Zip Code	
			DORVAL QU Street Address										
Operator of Aircraft			Same as Reg'd Aircraft Owner										
Same as Reg'd Aircraft Owner		(City							Sta	te	Zip Code	
Operator Does Business As: ROYAL AIRLINES Operator Designator Code: QN													
- Type of U.S. Certificate(s) Held: None													
Air Carrier Operating Certificate(s):													
Operating Certificate: Operator Certificate:													
Regulation Flight Conducted Under: Part 129: Foreign													
Type of Flight Operation Conducted: Non-se	cheduled; Inter	natio	nal; Pa	ssenger/C	argo								
	FACT	UAL	REPO	RT - AVI	ATION							Page 2	

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AVIATION				се Туре: Асс	cident							
First Pilot Information												
Name				City			State	Date of Birth	Age			
On File		On File			On File	On File	36					
Sex: M Seat Occupied	: Left	Pri	ncipal Profes	ı Pilot	Certificate Number: On File							
Certificate(s): Airline Transport												
Airplane Rating(s): Multi-engine Land; Multi-engine Sea; Single-engine Land; Single-engine Sea												
Rotorcraft/Glider/LTA: None												
Instrument Rating(s): Airplane												
Instructor Rating(s): None												
Type Rating/Endorsement for Accident/Incident Aircraft? Yes Current Biennial Flight Review?												
Medical Cert.: Class 1	Medica	al Cert. Statu	s: Valid Med	dicalno wa	ivers/lim.		Date of La	st Medical	Exam: 12/1997			
- Flight Time Matrix	All A/C	This Make and Model	Airplane Single Engine	Airplane Mult-Engine	Night	Ins:	rument Simulated	Rotorcra	ft Glider	Lighter Than Air		
Total Time	7900	2700	1200	6700	1200	50) 15)				
Pilot In Command(PIC)	3500	1000	1000	2500	300	25	7:	5				
Instructor	070	070		070	400		_					
Last 90 Days Last 30 Days	270 90	270 90		270 90	100 30	2	3	-				
Last 24 Hours	4	4		4	1	,	7					
Seatbelt Used? Yes			Used? Yes		Toxico	ology Perfo	rmed? No		Second Pilot? Y	es		
	I				I							
Flight Plan/Itinerary												
Type of Flight Plan Filed: IF	R											
Departure Point					State	Air	port Identifie	r Dep	arture Time	Time Zone		
Same as Accident/Incide	nt Location					FL	L	211	4	EST		
Destination					State	Air	port Identifie	ort Identifier				
TORONTO	ON											
Type of Clearance: IFR												
Type of Airspace: Class C												
Weather Information												
Source of Briefing: Company												
Method of Briefing:												
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Occurrence Type: Accident

	FITTBORE			Juli elice i	уре. /	Acciden	ι						
Weather Information													
WOF ID	Observation Time	Time Zone	WOF	Elevation		WOF Di	stance From	n Accid	dent Site		Direction From Accident Site		
FLL	2114	EST		11 Ft. MS	SL				1 NM		270 Deg. Mag.		
Sky/Lowes	t Cloud Condition: Scat				2	2000 Ft. AG	L	Condition o	f Ligh	ght: Night/Dark			
Lowest Ce	iling: None			0 Ft. AG	iL	Visibi	lity:	15	SM	Altii	meter:	30.00	"Hg
Temperatu	ıre: 28 °C	2	2 °C V	Vind D	Direction:	120			Dei	nsity Altitude:	1483	Ft.	
Wind Spee	ed: 12	Gusts:		V	Veathe	er Condtions at Accident Site: Visual Conditions							
Visibility (R	RVR): 0 Ft.	Visibility	(RVV)	0 8	SM	Intensity	of Precipita	ation: (Unknown				
Restrictions to Visibility: None													
Type of Precipitation: None													
Accident	Information												
Aircraft Damage: Minor Aircraft Fire: None Aircraft Explosion None													
Classificati	on: Foreign Registere	d/U.S. Soil											
- Injury Su	mmary Matrix	Fatal	Serious	Minor	N	None	TOTAL						
First Pi	lot					1	1						
Second	l Pilot					1	1						
Studen	t Pilot												
Flight I	nstructor												
Check	Pilot												
Flight E	ngineer					1	1						
Cabin A	Attendants					5	5						
Other C	Crew												
Passen	gers		3	1	4	169	186						
- TOTAL A	ABOARD -		3	1	4	177	194						
Other G	Ground	0	0		0		0						
- GRAND	TOTAL -	0	3	1	4	177	194						

National Transportation Safety Board

FACTUAL REPORT AVIATION

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Occurrence Type: Accident

Administrative Information

Investigator-In-Charge (IIC)

ALAN C. STONE

Additional Persons Participating in This Accident/Incident Investigation:

MICHAEL B BUSH FAA FSDO FT. LAUDERDALE, FL 33315

MARK CLITSOME TSB CANADA QUEBEC,

RICHARD S PARKER PRATT & WHITNEY EAST HARTFORD, CT 06108

MIKE MALKA ROYAL AIRLINES DORVAL,