Landing gear failure, Boeing 727-200, July 9, 1997

Micro-summary: The right main landing gear on this Boeing 727-200 had partially collapsed as the passengers were boarding.

Event Date: 1997-07-09 at 840 EST

Investigative Body: National Transportation Safety Board (NTSB), USA

Investigative Body's Web Site: http://www.ntsb.gov/

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National Transportation Safety Board	NTSB	ID: CHI97IA205	5	Aircraft Registration Number: N770AT				
FACTUAL REPORT	Occur	rence Date: 07/0	9/1997	Most Critical Injury: None				
AVIATION	Occur	rence Type: Incid	ent	Investigated By: NTSB				
Location/Time								
Nearest City/Place	State	Zip Code	Local Time	Time Zone				
INDIANAPOLIS	IN	46241	0840					
Airport Proximity: On Airport	Distance From	m Landing Facility:		Direction From Airport:				
Aircraft Information Summary								
Aircraft Manufacturer	Model/Serie	S		Type of Aircraft				
Boeing	B-727-200			Airplane				
Sightseeing Flight: No Air Medical Transport Flight: No								

Narrative

Brief narrative statement of facts, conditions and circumstances pertinent to the accident/incident:

On July 9, 1997, at 0840 eastern standard time, a Boeing 727-200, N770AT, operated by American Trans Air, was determined to have received minor damage prior to pushback from the gate. During his pre-flight walk around, the Flight Engineer discovered that the aft trunnion on the shock strut of the right main landing gear had failed. The Captain was notified of the discrepancy by the Flight Engineer. The Captain halted the boarding of passengers. The seven crewmembers and four passengers were deplaned by normal means. There were no injuries. The 14 CFR Part 121, Flight 403/402, was scheduled to depart Indianapolis International Airport, Indianapolis, Indiana, with Las Vegas International Airport, Las Vegas, Nevada as the intended destination. Instrument meteorological conditions prevailed and an instrument flight plan had been filed.

The aircraft had landed at Indianapolis International Airport the night before and had received routine maintenance. The next morning the aircraft was towed to gate C-4 at the main terminal. The maintenance personnel who serviced the aircraft and towed the airplane to the gate reported that they did not notice anything unusual about the right wing or the right main landing gear of the airplane. The aircraft fueler reported that he noticed that the right wing was lower than the left wing. He notified maintenance personnel of the condition. About the same time, the Flight Engineer discovered and reported the condition of the right main landing gear to the maintenance personnel. It was determined that the outer cylinder trunnion had fractured and the actuator support beam was resting on the fractured stub of the outer cylinder trunnion. The airplane was towed to a hangar for removal and replacement of the shock strut and local repair of the main landing gear support beam.

It was determined that the fractured right main landing gear, p/n 65-17650-74, s/n 0171601597, had recently been overhauled and had accumulated 51 cycles and 126 hours since installation 21 days prior to the fracture. The aft trunnion of the right main landing gear and the trunnion bearing components were sent to the Materials Laboratory of the National Transportation Safety Board for examination.

The examination revealed that the aft portion of the trunnion was fractured into four pieces with the three largest pieces contained within a cylindrical repair sleeve encircling the trunnion. The repair sleeve, installed during a previous repair of the outer cylinder trunnion journal, was also longitudinally split at the approximate bottom centerline of the trunnion.

An etched longitudinal metallographic section was cut through the initiation area of the upper aft lug fracture. The metallographic section uncovered a thin layer of plating on the outer diameter surface of the lug. The plating was determined to be chromium by energy dipersive spectrographic analysis. The plating extended from the fracture location aft through the transition radius and onto the outer diameter of the lug. The plating was thickest on the lug diameter (0.0022 inch), thinned in the radius, and at an intermediate thickness (0.00035) adjacent to the fracture.

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Narrative (Continued)

Another section was cut from the lower lug showing a portion of the journal diameter, the trunnion end face and the transition radius from the end face to the lug outer diameter. Chromium plating was apparent on the entire manufactured surface visible in this section. On the journal diameter the plating thickness measured 0.005 inch. The plating extended aft over the corner chamfer onto the end face and throughout the transition radius.

The Boeing Commercial Airplane Overhaul Manual indicated that chromium plating was only to be applied to the trunnion journal outer diameter and a portion of the end face. It also specifically denoted areas for chromium plate runout that bound these areas. Further, the figure had a "No Chrome" notation for the journal surface immediately adjacent to the chamfer. The landing gear overhaul facility's "workorder traveler" for the gear indicated similar chromium plating details. Flag note 13 in figure 406 indicated plating the aft lugs and surrounding areas with either titanium-cadmium or low hydrogen embrittlement (LHE) cadmium. Work order documents supplied by the overhaul facility indicated that LHE cadmium was used.

In addition to the trunnion section of the landing gear, the spherical bearing assembly, locking bolt and anti-rotation washer were received. For normal assembly, the aft trunnion of the gear is inserted into the inner diameter of the wing mounted spherical bearing.

The spherical bearing that contacts the trunnion journal had an inner diameter which measured approximately 3.50 inches. The aft trunnion journal of the failed landing gear had an outer diameter that measured approximately 3.25 inches. During landing gear replacement, the maintenance procedures require that the proper sized spherical bearings, either 3.50 inch or the 3.25 inch inner diameter, be matched with the appropriate outer cylinder aft trunnion outside diameter. (See Metallurgist's Factual Report)

Boeing had issued Service Bulletins in 1980 and 1991 which detailed the requirements to ensure that the correct trunnions and spherical bearings were matched during landing gear replacement.

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VETY BOX		Occur	currence Type: Incident										
Landing Facility/Approach In	formation		A: ID	Τ	. = 1				Ī	1	T	D	\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \
Airport Name		- 1	Airport ID:	Aii	rport Elevatio			way Used	Runwa	ay Length		Runv	vay Width
INDIANAPOLIS INTL			IND		Ft. N	VISL	0						
Runway Surface Type:													
Runway Surface Condition:		_	_		_	_	_			_	_		_
Type Instrument Approach:													
VED Approach/Landing:													
VFR Approach/Landing:													
Aircraft Information													
Aircraft Manufacturer			Mode	el/Seri	ies					Serial	Numbe	er	
Boeing			B-72	27-20	00					2195	3		
Airworthiness Certificate(s): Transport													
Landing Gear Type: Retractable - Tricycle													
Homebuilt Aircraft? No	Number of Seats: '		Certified Max Gross Wt. 197700								of Engines: 3		
				Engine Manufacturer: Model/Series: JT8D-17									ed Power: 000 LBS
- Aircraft Inspection Information													
Type of Last Inspection			Date of La	Date of Last Inspection Time Sin				nce Last Insp		Airfrar	ne To	tal Time	
Continuous Airworthiness			06/1997	06/1997 162 Hou						ours	urs 52509 Hours		
- Emergency Locator Transmitter (ELT) Information												
ELT Installed? Yes	ELT Operate	ed? No		ELT Aided in Locating Accident Site?									
Owner/Operator Information													
Registered Aircraft Owner			Street	Addre									
AMERICAN TRANS AIR			City	7337 WEST WASHINGTON ST.									Zip Code
			City INDIANAPOLIS									.6	46251
			Street Address										
Operator of Aircraft				Same as Reg'd Aircraft Owner								. 1	7: 0 1
Same as Reg'd Aircraft Owner	City	City							Stat	e	Zip Code		
Operator Does Business As:							Op	perator Desig	nator Co	ode:			
- Type of U.S. Certificate(s) Held:							-						
Air Carrier Operating Certificate(s):	: Flag Carrier/Dom	nestic											
Operating Certificate:					Operator Ce	ertific	ate:						
Regulation Flight Conducted Under	r: Part 121: Air Ca	arrier											
Type of Flight Operation Conducted	d: Non-scheduled	; Dome	estic; Pass	senge	er Only								
		EA CTI	IAL DED	ОВТ	AMATIC	N.I							Dogo 2
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	AVIATI	ON		Occurren	ce Type: Ind	cident								
First Pilot	t Information													
Name						City				State	Date of Birth	Age		
On File						On File				On File	On File	51		
Sex: M	Seat Occupied	: Left	Prin	cipal Profes	sion: Civilia	n Pilot			Certi	ficate Num	ber: On File			
Certificate(s): Airlir	ne Transpor		•										
Airplane Ra	ating(s): Multi	i-engine Lar	nd; Single-ei	ngine Land										
Rotorcraft/Glider/LTA: None														
Instrument Rating(s): Airplane														
Instructor Rating(s): None														
Type Rating/Endorsement for Accident/Incident Aircraft? Yes Current Biennial Flight Review?										eview?				
Medical Ce	ert.: Class 1	Medica	al Cert. Status	: Valid Me	dicalno w	aivers/lir	n.	Dat	e of Las	st Medical	Exam: 02/1997			
		•						'						
- Flight Tim	ight Time Matrix All A/C This Make and Model		Airplane Single Engine	Airplane Mult-Engine	Night	Actua	Instrument S	mulated	Rotorcraft	Glider	Lighter Than Air			
Total Time		6500												
Pilot In Cor	mmand(PIC)													
Instructor														
Last 90 Da	ys	160												
Last 30 Day						 				-				
Last 24 Ho	urs					1 -								
Seatbelt Us	sed? No	Shou	lder Harness	Used? No		Т	oxicology P	Performed? No Second Pilot? No						
Flight Pla	ın/Itinerary													
	ght Plan Filed: IF	 R												
Departure I							State	Airport Id	lentifier	Depa	arture Time	Time Zone		
	Accident/Incide	nt Location						IND		1825		EDT		
Destination	1						State	Airport Id	lentifier					
LAS VEG	iAS						1V	LAS						
Type of Cle	earance: None									•				
Type of Air	space:													
Weather	Information													
Source of	Briefing:													
Method of	Briefing:													
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AVIATION				Occurrence Type: Incident										
Weather	Information													
WOF ID	Observation Time	Time Zone	WOF	Elevation		WOF Distance From Accident Site					Direction From	n Accident Si	te	
	0000			0 Ft. MS	SL				0 NM		0 Deg. Mag.			
Sky/Lowes	st Cloud Condition: Unkr	nown		0 Ft. AGL						Condition of Light: Day				
Lowest Ce	eiling: Overcast		3	00 Ft. AG	iL	Visibi	Visibility: 1 SM			Alti	meter:	30.00	"Hg	
Temperatu	ure: 17 °C		16 °C Wind Direction: 360							nsity Altitude:		Ft.		
Wind Spee	Wind Speed: 11 Gusts: Weather Conditions at Accident Site: Instrum								ite: Instrume	ent C	Conditions			
Visibility (R	RVR): 0 Ft.	Visibility	(RVV)	0 8	SM	Intensity	y of Precipita	ation: I	Unknown					
Restrictions to Visibility: None														
Type of Precipitation: Drizzle														
Accident	Information													
Aircraft Dar	mage: Minor	Aircr	raft Fire: N	None				Aircraft Exp	losio	n None				
Classificati	ion: U.S. Registered/U	.S. Soil												
- Injury Su	mmary Matrix	Fatal	Serious	Minor	1	None	TOTAL							
First Pi	ilot					1	1							
Second	d Pilot					1	1							
Studen	nt Pilot													
Flight I	nstructor													
Check	Pilot													
Flight E	Engineer					1	1							
Cabin /	Attendants					4	4							
Other C	Crew													
Passen	ngers				\top	4	4							
- TOTAL A	ABOARD -					11	11							
Other G	Ground	0	0		0		0							
- GRANE	D TOTAL -	0	0		0	11	11							
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National Transportation Safety Board

FACTUAL REPORT AVIATION

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Occurrence Date: 07/09/1997

Occurrence Type: Incident

Administrative Information

Investigator-In-Charge (IIC)

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