Runway overrun following brake failure, Boeing 737-247, January 19, 1995

Micro-summary: This 737-247 overran the runway on landing following a loss of wheel brakes.

Event Date: 1995-01-19 at 0940 EST

Investigative Body: National Transportation Safety Board (NTSB), USA

Investigative Body's Web Site: http://www.ntsb.gov/

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NTSB ID: ATL95IA043

Aircraft Registration Number: N4515W

Occurrence Date: 01/19/1995

Most Critical Injury: None

Occurrence Type: Incident

Investigated By: NTSB

Location/Time

Nearest City/Place
ATLANTA

State
GA

Zip Code
30320

Distance From Landing Facility: 1

Direction From Airport: 40

Aircraft Information Summary

Aircraft Manufacturer Model/Series Type of Aircraft
BOEING 737-247 Airplane

Sightseeing Flight: No Air Medical Transport Flight: No

Narrative

Brief narrative statement of facts, conditions and circumstances pertinent to the accident/incident:

HISTORY OF FLIGHT

On January 19, 1995, at 0940 eastern standard time, a Boeing 737-247, N4515W, overran runway 9R at the Hartsfield Atlanta International Airport, in Atlanta, Georgia. There were no injuries to the airline transport captain and first officer, the three flight attendants, and the 23 passengers. The aircraft had minor damage. Flight 312 was operated under the provisions of 14 CFR Part 121 by Air South, Inc. Instrument meteorological conditions existed at the time, and an instrument flight rules flight plan was in effect for the scheduled, domestic, passenger flight. The flight originated in Columbia, South Carolina at 0850.

The captain stated the following: the engine start, taxi, and takeoff from Columbia were "ops normal." There were no unusual aircraft logbook entries; nothing pertaining to the brakes, anti-skid system, hydraulics, etc. The brakes seemed to perform normally on the ground in Columbia. There were no abnormal gauge indications, nor any warning or caution lights during the entire flight. The instrument landing system (ILS) approach and touchdown to runway 9R were normal. The first officer was the flying pilot. The ILS was manually flown, and the aircraft broke out of the weather at about 500 feet above ground level. The computed approach speed was in the mid-130 knot range. Following a normal approach (no glide path deviations were noted), the aircraft touched down about 1,500 to 2,000 feet down the runway, about 120 knots. The first officer reported that he began to apply wheel brakes about 80 to 90 knots. He remarked to the captain that the anti-skid system seemed to be "releasing." Thrust reverser operation was normal in all respects. The captain took control of the airplane, and checked the brakes. Brake pedal pressure was initially felt, then the brake pedals could be pressed all the way to the floor, and no braking action was observed. The aircraft departed the paved surface at the departure end of the runway. The aircraft came to a stop on the grass, about 200 feet beyond the edge of the departure end threshold.

The captain stated that there were never any cockpit indications indicating brake or hydraulic system problems during the landing roll. The captain did not feel that the aircraft had been hydroplaning, and in his opinion, the runway was dry. The first officer described the runway as having "damp spots."

PERSONNEL INFORMATION

Information of the captain is contained in this report at the section titled "First Pilot Information." Copies of his training records are included as an attachment to this report.

Information on the first officer is contained in this report at Supplement E. Copies of his training records are included as an attachment to this report.

AIRCRAFT INFORMATION

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Narrative (Continued)

According to Air South personnel, the aircraft had been recently acquired, and had been in operation for about 35 hours (44 cycles) since Air South's acquisition.

The aircraft underwent maintenance work at AAR Oklahoma, Inc. prior to its acceptance by Air South. According to non-routine forms provided by the operator, on January 5, 1995, a ground test of the anti-skid control system was performed in accordance with the Boeing 737 Maintenance Manual Chapter 32-42-0. On January 6, 1995, the left hand inboard and outboard anti-skid control valves were found to be inoperative. They were replaced and operationally checked. According to maintenance records, the only subsequent work performed on the anti-skid system occurred on January 11, 1995. On that day, the left main landing gear anti-skid harness was resecured, following a writeup indicating that it needed to be secured. This work was performed by the operator.

METEOROLOGICAL INFORMATION

Instrument meteorological conditions existed at the time of the incident. Additional information is located at the section titled "Weather Information."

FLIGHT RECORDERS

The cockpit voice recorder (CVR) and digital flight data recorder (DFDR) were secured following the incident, and forwarded to the NTSB Headquarters in Washington, DC for examination.

An initial readout of the CVR revealed that the unit had not been disabled by the flight crew, thus it continued operating after the aircraft had come to a stop. No useful information was obtained from the recording.

The DFDR data for this incident indicate that the airplane touched down at 132 knots indicated airspeed (IAS), at a pressure altitude of 931 feet, and a magnetic heading of 95 degrees. The data indicate two possible times where the aircraft overran the runway. Assuming a runway length of 9,000 feet, the DFDR data indicated that the airplane touched down either 2,501 or 2,273 feet from the runway 9R approach end. For additional information, refer to the "Digital Flight Data Recorder Study", attached to this report.

WRECKAGE AND IMPACT INFORMATION

The only verified damage to the aircraft was a blown main tire on the left, outboard position. The other three main tires remained intact and inflated, except there was evidence of skidding on the right, outboard tire. The tread of the right, outboard tire showed evidence of rubber reversion. There were skid marks on the runway surface, leading to the tracks in the mud made by the aircraft. An examination of the track which led to the right, outboard tire showed that it had a "steam cleaned" appearance, i.e. it was lighter in color than the surrounding runway surface.

TESTS AND RESEARCH

Following the incident, several tests and examinations were performed. An inspection of the aircraft after the incident revealed that the left, inboard anti-skid valve brake and return lines were crossed. Also, after the aircraft had been returned to service, abnormal main tire wear was observed at the left, outboard position. Further inspection revealed that the wiring to the left, inboard and left, outboard wheel speed transducers were crossed. This discrepancy was corrected, and no further reports of abnormal tire wear were received, or observed.

All four main tires, as well as their respective wheel and brake assemblies, were shipped to Thompson Aerospace, Inc., in Miami, Florida. A functional test of each brake assembly was conducted; all operated normally. Brake wear was not greater than 50 percent on any brake assembly.

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Narrative (Continued)

The anti-skid control unit, and the number two anti-skid control valve were removed from the incident aircraft. These units were shipped to Aircraft Braking Systems Corporation for testing and examination. Functional testing of the anti-skid valve revealed that it was functional. Testing of the anti-skid control unit (box) revealed that the left, inboard skid detect circuit was inoperative, due to an open Q24 transistor.

According to the Boeing Commercial Airplane Group, the discrepancies discovered would result in the following:

- (1) If the left, outboard wheel started to skid, the wheel speed transducer signal would go to the left, inboard wheel skid detector circuit, which was inoperative. This would have allowed tire damage to occur to the left, outboard tire, since there would be no brake release.
- (2) If the left, outboard wheel locked up, the wheel speed transducer signal would go to the left, inboard wheel detector, providing a release of the left, inboard wheel, if the inboard locked wheel protection circuit was armed. Arming of the locked wheel protection circuit occurs when either the inboard wheel spin up to more than 30 mph, or the air/ground sensor switch is in the "air" position, to provide touchdown protection.
- (3) With the left inboard valve hydraulic lines crossed, sluggish braking would occur on the left, inboard wheel.

The component examinations did not reveal evidence of malfunction of the right, main gear anti-skid or braking systems. The investigation was not able to determine if the crossed brake and return lines on the left, inboard anti-skid valve had any effect on right side braking.

Boeing Commercial Airplane Group was asked to provide information to determine if the aircraft inspections performed by AAR Oklahoma, Inc. should have uncovered the discrepancies found in N4515W. Regarding the inoperative Q24 transistor, the ground test of the anti-skid system may have been affected by the crossed wheel speed transducer wiring. Without additional testing, it could not be determined if the inoperative Q24 transistor would have been discovered. Regarding the crossed wheel speed transducer wiring, the appropriate test for this was not referenced in the non-routine work cards, however that test is required if the landing gear is replaced. Regarding the crossed anti-skid valve brake and return lines, Boeing reported that this condition would result in sluggish braking on the left, inboard wheel.

ADDITIONAL INFORMATION

The tested aircraft system components were released to the operator, Air South. Inc.

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AVIATION			rrence Typ	e: Incident								
Landing Facility/Approach Information												
Airport Name	Airport ID:	Airport Elevation Runway Used		way Used	Runwa	ay Lengt	h R	ınway Width				
HARTSFIELD ATLANTA INTL.			ATL	150 Ft.	MSL	9R	9000		00 15		50	
Runway Surface Type: Concrete												
Runway Surface Condition:												
Type Instrument Approach: ILS-complete												
VFR Approach/Landing: None												
Aircraft Information												
Aircraft Manufacturer BOEING				el/Series -247						Serial Number 19612		
Airworthiness Certificate(s): Transport												
Landing Gear Type: Retractable - Tricycle												
Homebuilt Aircraft? No	Number of Seats:								Number of Engines		es: 2	
				Engine Manufacturer: Model/Serie: P&W JT8D-9A							ated Power: 4500 LBS	
- Aircraft Inspection Information												
Type of Last Inspection	Date of Last Inspection Time Sir				nce Last Inspection			Airframe Total Time				
Continuous Airworthiness				01/1995			2 Hours			37421 Hours		
- Emergency Locator Transmitter (ELT) Information												
ELT Installed?	T Installed? ELT Operated? ELT Aided in Locating Accident Site?											
Owner/Operator Information												
Registered Aircraft Owner Street Address 600 SUNSET RIDGE												
MIMI LEASING CORP.				City							Zip Code 52003	
	DUBUQUE IA 52003 Street Address											
Operator of Aircraft P.O. BOX 11129												
AIR SOUTH, INC.	City							State SC	Zip Code 29211			
Operator Does Business As:	-			O	perator Desig	nator Co	ode: A6	XA				
- Type of U.S. Certificate(s) Held:												
Air Carrier Operating Certificate(s)	: Flag Carrier/Dom	nestic										
Operating Certificate: Operator Certificate:												
Regulation Flight Conducted Under: Part 121: Air Carrier												
Type of Flight Operation Conducted	d: Scheduled; Dor	nestic;	Passenge	er Only								
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TACIDAL REPORT			/13/133		—				
AVIATION	Occurrence Type: Incident								
First Pilot Information									
Name		City				State	Date of Birth	Age	
On File		On File)			On File	On File	34	
Sex: M Seat Occupied: Left Princ	cipal Profess	ın Pilot			Certi	ificate Nun	nber: On File		
Certificate(s): Airline Transport; Flight Engi									
Airplane Rating(s): Multi-engine Land									
Rotorcraft/Glider/LTA: None									
Instrument Rating(s): Airplane									
Instructor Rating(s): None									
Type Rating/Endorsement for Accident/Incident Aircraft	? Yes			Current	Biennial F	light Re	eview?		
Medical Cert.: Class 1 Medical Cert. Status:	Valid Med	dicalno w	aivers/li	n.	Dat	e of Las	st Medical	Exam: 01/1995	
- Flight Time Matrix All A/C This Make and Model	Airplane Single Engine	Airplane Mult-Engine	Nigh	Actu	Instrument Simulat		Rotorcraft	Glider	Lighter Than Air
Total Time 3634 761									
Pilot In Command(PIC)									
Instructor									
Last 90 Days 161 161									
Last 24 Hours 8 8			1				<u> </u>		
Seatbelt Used? Yes Shoulder Harness I	Used? Yes		Toxicology Performed? No Second Pilot? Yes						
Flight Plan/Itinerary									
Type of Flight Plan Filed: IFR									
Departure Point		Т	State Airpor		port Identifier		arture Time	Time Zone	
COLUMBIA			SC CAE					EST	
Destination			State Airport		port Identifier				
Same as Accident/Incident Location		ATL							
Type of Clearance: IFR									
Type of Airspace:									
Weather Information									
Source of Briefing: Company									
Method of Briefing:									
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AVIATION			Oc	Occurrence Type: Incident										
Weather Information														
WOF ID	Observation Time	vation Time			WOF Elevation WOF Distance			ance From Accident Site			Direction From Accident Site			
ATL	1028	EST	1(026 Ft. N	/ISL				0 NM		0 Deg. Mag.			
Sky/Lowes	st Cloud Condition: Unkr	nown		0 Ft. AGL					Condition of Light: Day					
Lowest Ceiling: Overcast				300 Ft. AGL Visibility: 2				2	SM	SM Altimeter: 29.00			"Hg	
Temperatu	nperature: °C Dew Point: °C Wind Direction: 1					120			Dei	nsity Altitude:		Ft.		
Wind Spee	Wind Speed: 13 Gusts:					ner Condti	ions at Accid	dent Si	Site: Instrument Conditions					
Visibility (R	sibility (RVR): 0 Ft. Visibility (RVV) 0 SM Intensity of Precipi						of Precipita	ation: I	iion: Unknown					
Restriction	ns to Visibility: Fog											<u> </u>		
Type of Precipitation: None														
Accident Information														
Aircraft Damage: Minor				Aircraft Fire: None					Aircraft Exp	losio	n None			
Classification: U.S. Registered/U.S. Soil														
- Injury Su	mmary Matrix	Fatal	Serious	erious Minor			TOTAL							
First Pil	lot					1	1							
Second	d Pilot					1	1							
Studen	nt Pilot													
Flight In	nstructor													
Check I	Pilot													
Flight E	Engineer													
Cabin A	Attendants					3	3							
Other C	Crew													
Passen	Passengers					23	23							
- TOTAL A	ABOARD -					28	28							
Other C	Other Ground 0			0 0 0										
- GRAND	O TOTAL -	0	0		0	0 28 28								

National Transportation Safety Board

FACTUAL REPORT AVIATION

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Occurrence Type: Incident

Administrative	Information

Investigator-In-Charge (IIC)

RALPH E. HICKS

Additional Persons Participating in This Accident/Incident Investigation:

DAVID A DEES FAA/GEORGIA FSDO COLLEGE PARK, GA 30337

CASSANDRA L JOHNSON NTSB/RE-20 WASHINGTON, DC 20594